



A novel deep unsupervised learning-based framework for optimization of truss structures

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Abstract

In this paper, an efficient deep unsupervised learning (DUL)-based framework is proposed to directly perform the design optimization of truss structures under multiple constraints for the first time. Herein, the members' cross-sectional areas are parameterized using a deep neural network (DNN) with the middle spatial coordinates of truss elements as input data. The parameters of the network, including weights and biases, are regarded as decision variables of the structural optimization problem, instead of the member's cross-sectional areas as those of traditional optimization algorithms. A new loss function of the network model is constructed with the aim of minimizing the total structure weight so that all constraints of the optimization problem via unsupervised learning are satisfied. To achieve the optimal parameters, the proposed model is trained to minimize the loss function by a combination of the standard gradient optimizer and backpropagation algorithm. As soon as the learning process ends, the optimum weight of truss structures is indicated without utilizing any other time-consuming metaheuristic algorithms. Several illustrative examples are investigated to demonstrate the efficiency of the proposed framework in requiring much lower computational cost against other conventional methods, yet still providing high-quality optimal solutions.

Keywords Unsupervised learning · Deep neural network · Loss function · Truss optimization

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1 Introduction

Size optimization of truss structures has attracted the considerable attention of many scholars during the last decades. As known, this problem often minimizes the structural weight to mandatorily satisfy multiple constraints. As indicated in Refs. [1–3], the constraints of such problems are non-convex and highly nonlinear functions. There have been a variety of algorithms proposed for solving this issue, and they are usually categorized into two types: gradient and gradient-free. The first one relies on the gradient information to search optimum solutions. For instance, an optimality criterion was developed by Khot et al. [4, 5] to minimize the weight of the truss structure. El-Sayed et al. [6] described a combining model using linear and nonlinear goal programming. Besides, the optimality criteria integrated with the nonlinear analysis technique were delivered by Saka and Ulker [7]. Although these methods have achieved certain successes, they often encounter several challenges as indicated in Ref. [8]. In specific, it can fail due to the unavailable gradient information of the objective and constraints functions. Among different approaches, evolutionary and

population-based metaheuristic algorithms have been extensively applied to solve optimization problems for truss structures, such as improved differential evolution (IDE) [2], firefly algorithm (FA) [9], adaptive hybrid evolutionary FA (AHEFA) [10], particle swarm algorithm (PSO) [11], democratic PSO (DPSO) [12], genetic algorithms (GAs) [13], hybrid optimality criterion and genetic algorithm (OC-GA) [14], modified coyote optimization algorithm (MCOA) [15], harmony search (HS) [16], evolutionary ant colony optimization (EACO) [17], improved grey wolf optimizer (IGWO) [18], efficient HS (EHS) [19], and balancing composite motion [20], etc. Despite these algorithms can achieve near-global optimum solutions, they also suffer from several drawbacks, such as high computational cost, slow convergence speed, and many turning parameters [21].

Over the past decade, machine learning (ML) has demonstrated to be effective and promising in many fields to assist decision-making in image recognition [22], self-driving systems [23], medical diagnoses [24], financial services [25], structural analysis [26, 27], to name a few. Indeed, it has displayed salient advantages over conventional methods in handling complex practical problems that do not have a closed-form expression. Recently, DNN which is one of the most widely used ML models has attracted remarkable attention in computational mechanics, for example, structural analysis [28, 29], structural health monitoring [30–33], materials sciences [34–36], and structural optimization is no exception. Since the 1990s, a neural network (NN) combined with other algorithms was developed for optimization of truss structures [37–42]. More recently, Mai et al. [43] proposed an integrated model of DNN and differential evolution (DE) algorithm for optimization of truss structures with geometrically nonlinear behavior. The data-driven ML approach was introduced by Long [44] for solving structural optimization problems. In addition, DNN integrated with the traditional algorithms for tackling multi-objective optimization problems [45, 46]. Besides, it has also been successfully applied to solve topology optimization problems of structures. For instance, Li et al. [47] developed an integrated pre-trained network to replace the iterative optimization for designing heat conduction structures. A surrogate model based on NN was proposed by White et al. [48] to perform optimization of macroscale elastic structures. Also, Deng and Albert [49] used DNN to approximate an implicit function of the level set for topology optimization. To decrease computational cost, Abueidda et al. [50] employed a convolutional neural network (CNN) model to optimize structures with geometric nonlinearity. Despite the remarkable success achieved in this area, it is worth mentioning that most of the above studies use a DNN as a surrogate model, and employ supervised learning to construct the model. The above approaches have several disadvantages, making them inefficient such as: (i) they require a time-consuming effort due to the numerical

simulation for collecting data of training process; (ii) it is difficult to estimate the suitable training data size, and (iii) they depend strongly on the quality and quantity of data to build the high-accuracy data-driven predictable model. To circumvent these challenges, physics-informed machine learning frameworks, which integrated the governing physics law into the unsupervised learning process, have obtained increasing attention and successfully applied in many tasks, such as structural analysis [51–53], solving partial differential equations (PDEs) [54, 55], and fluid mechanics [56, 57]. For the structural optimization problem, Chandrasekhar and Suresh [58] firstly proposed a direct topology optimization paradigm using unsupervised learning NN. However, in this context, it remains a bottleneck in the computational effort that a part of the sensitivity analysis is performed by direct differentiation. Furthermore, to the best of our knowledge, a multilayer direct DNN model has still not been yet utilized for optimization of truss structure thus far.

As another alternative effective approach for the truss optimization field, this study aims to firstly propose an efficient DNN-based framework using unsupervised learning for optimization of truss structures under multiple constraints. Our work is different from most existing publications based on ML in a number of ways. In the proposed framework, the training data of the network is a set of the central spatial coordinates of all truss members which can be easily collected from the connectivity information of the structure without any numerical simulations, independent sampling techniques, as well as its small size. In addition, the design variables of truss optimization are the weights and biases of the DNN, instead of the cross-sectional areas of truss members. Therein, the sensitivity of loss function to parameters can be easily estimated by an automatic differentiation framework JAX [59] and backpropagation algorithm. And finally, the FEA is only required to assist in building the loss function while optimizer and backpropagation algorithm allow the network to directly participate in the process of structure optimization design. According to that, the optimum weight of truss structures can be indicated as soon as the training process ends with low computational overhead and without utilizing any other algorithms. The performance and applicability of the proposed approach are demonstrated through several benchmark truss optimization examples under various constraints. The outcomes of the proposed DUL are compared with conventional algorithms to evaluate its efficiency and reliability. The experimental results showed that DUL is able to dramatically reduce the computational cost, yet still yield high-quality optimal solutions.

The rest of the paper is organized as follows. The formulation of the size optimization problem of truss structures with multiple constraints is given in Sect. 2. Afterward, the efficient DNN-based structural optimization paradigm is presented in Sect. 3. Next, several numerical examples are

investigated to demonstrate the efficiency of the proposed method in Sect. 4. Finally, conclusions and challenges are outlined in Sect. 5.

2 Sizing optimization

For the size optimization of truss structures, the goal is to minimize the structural mass while still satisfying all design constraints. In which, the cross-sectional areas of truss members are considered as continuous design variables and predefined in a feasible region. The mathematical formulation of this problem can be represented as follows

$$\begin{aligned} \text{Minimize} \quad & W(\mathbf{A}) = \sum_{i=1}^m \rho_i A_i L_i, \\ \text{subjected to} \quad & g_j(\mathbf{A}) = \frac{q_j}{[q_j]} - 1 \leq 0, \quad j = 1, 2, \dots, n, \\ & A_k^{\text{low}} \leq A_k \leq A_k^{\text{up}}, \quad k = 1, 2, \dots, m, \end{aligned} \tag{1}$$

where \mathbf{A} is the design variable vector; A_k is the cross-section area of the k th member; $W(\cdot)$ is the mass of the truss structure; L_i and ρ_i are the length and material density of the i th member; g_j is the j th constraint function; q_j denotes the j th displacement, stress or frequency; $[q_j]$ is the j th allowable displacement, stress or frequency; m and n are the number of members and constraints, respectively; A_k^{low} and A_k^{up} are the lower and upper bound of A_k , respectively.

To solve the above optimization problem, a penalty function is used to convert the constrained optimization problem into an unconstrained optimization one [2, 10, 60]. Thus, Eq. (1) can be rewritten as follows

$$\begin{aligned} \text{Minimize} \quad & f(\mathbf{A}) = (1 + \varepsilon_1 c)^{\varepsilon_2} W(\mathbf{A}), \\ & c = \sum_{j=1}^n \max(0, g_j(\mathbf{A})), \end{aligned} \tag{2}$$

where c is the sum of the violated constraints of the design problem; ε_1 and ε_2 are parameters whose values control the exploration and exploitation rates of the design space.

In this work, the following two types of structural optimization problems are addressed: (1) frequency constraints only, and (2) displacement and stress constraints. For the first case, according to Kaveh [61, 62], ε_1 is set to 1 for a better control on other parameters. Note that the first part of Eq. (2) is a penalty function. And its value increases as the exponent ε_2 increases for infeasible designs. Hence, the coefficient ε_2 is set in a way that the penalty decreases. As mentioned by Kaveh [62], this coefficient directly effects the exploration of the algorithm, and its value varies from 1.5 to 3 of the search process. In this study, the value of ε_2 is set to 1.5 at the first epoch and linearly increased by 0.05 in each epoch until it achieves 3 [10, 12] in solving optimization problem under frequency constraints. For the remaining

case, a self-adaptive parameters scheme which is proposed by Sonmez [63] and Hasancebi [64] is utilized for the rest of experiments. Contrary to the first case, ε_2 is set to 1, while the coefficient ε_1 is automatically adjust according to the feedback from previous solution, as expressed below

$$\varepsilon_1^{(t)} = \begin{cases} (1/\kappa)\varepsilon_1^{(t-1)} & \text{if } f^{(t-1)} \text{ is feasible,} \\ \kappa \varepsilon_1^{(t-1)} & \text{if } f^{(t-1)} \text{ is infeasible,} \end{cases} \tag{3}$$

in which $\varepsilon_1^{(t)}$ and $\varepsilon_1^{(t-1)}$ denote the penalty coefficients at epochs t and $(t - 1)$, respectively. ε_1 is set to 1 at the beginning of the epoch. κ is the learning parameter of $\varepsilon_1^{(t)}$, and it is determined by the following equation [63, 64]

$$\kappa = 1 + \frac{1}{n} > 1.01, \tag{4}$$

where n is the total number of constraints.

3 Deep unsupervised learning-based optimization framework

The unconstrained optimization problem presented in Eq. (2) is usually solved by population-based search algorithms as indicated in Introduction. However, they often require a large number of FEA runs to achieve a good solution. In this section, an alternative approach based on DNN is suggested to directly perform optimization of truss structures with lowest computational cost. The overall flowchart of the proposed framework is illustrated in Fig. 1. Here, the weights and biases of the network θ are design variables. In the context of this paradigm, the initial weight and bias values are randomly chosen according to a normal distribution on interval $[-1, 1]$ for the network setup. The middle spatial coordinates of truss members are treated as the input data, while the cross-sectional areas are represented through parameters and defined as the outputs of the network $\hat{\mathbf{A}}$ obtained by the feedforward phase. Then, the loss function including the objective and constraints is established based on these outputs, and the responses of structures are attained from FEA. JAX, backpropagation, and optimizer are employed to estimate the sensitivity of the loss function to the parameters. To carry out the training, all of the above operations are going on repetitively until the loss function value reaches the minimum corresponding to the optimum weight of truss structures. In general, the proposed approach consists of the following three major components: (i) training data; (ii) DNN, and (iii) the loss function. They are described in detail in the following three subsections.

3.1 Training data

It should be noted that this work relies on unsupervised learning, where the model only has input data, and no corresponding

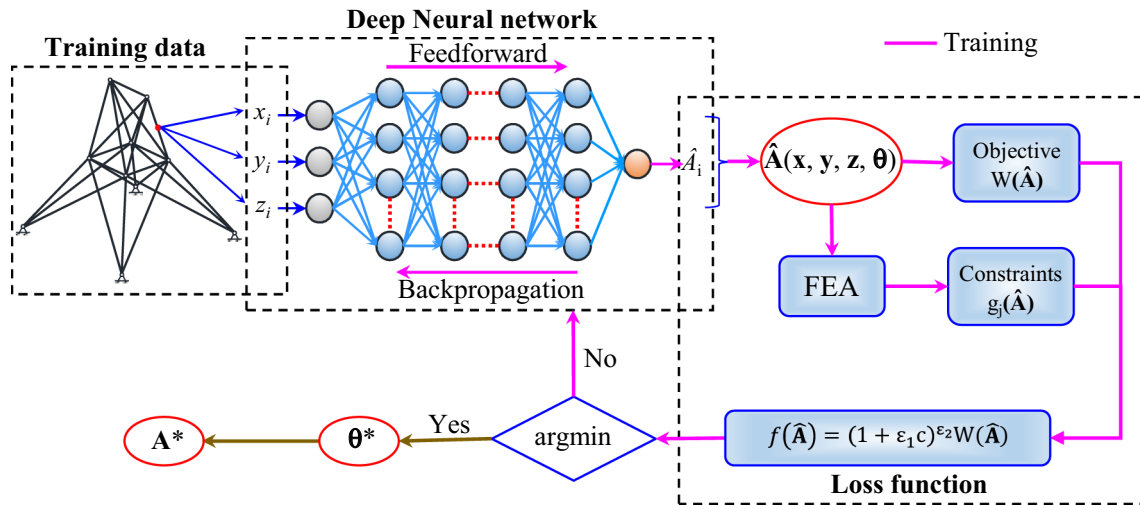


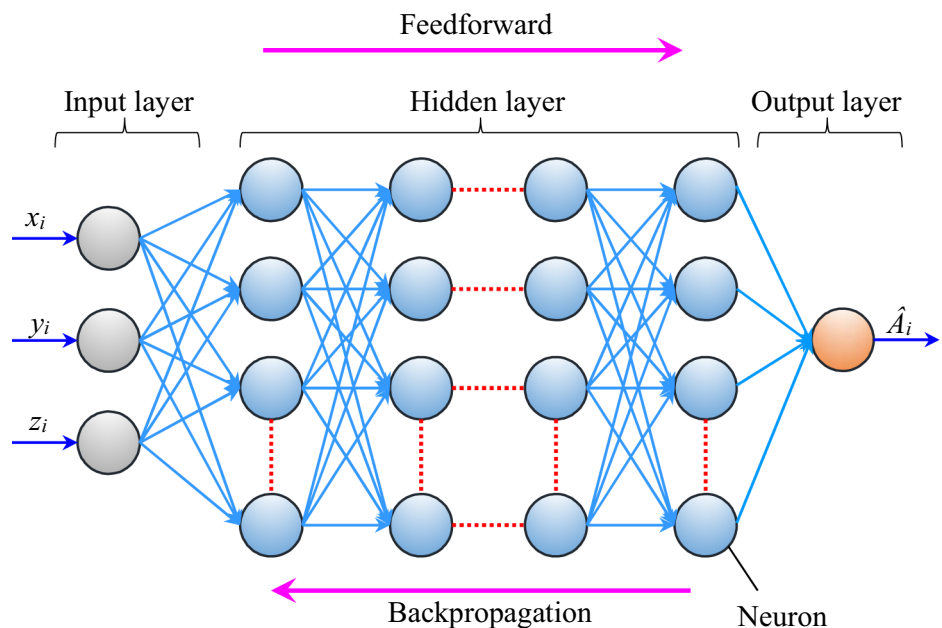
Fig. 1 Flowchart of the DUL-based framework for design optimization

output values are required. Furthermore, the input data only contains the information to describe the problem, such as geometric, material, loads, boundary conditions, and so on. In this work, a set of middle spatial coordinates (x_i, y_i, z_i) of truss elements is selected as training data. Clearly, they are simple and can be easily obtained from known geometry of the structure. Additionally, its size is small $(d \times 2)$ and $(d \times 3)$ for planar and space truss structures, respectively. Here d denotes the number of truss members, while 2 or 3 implies the number of spatial coordinates for 2- or 3-dimensional truss corresponding to the number of neurons in the input layer. Hence, it helps to reduce the complexity of NN as well as improve the convergence speed.

3.2 Deep neural network

The DNN, which is one of the ML models, is a set of mathematical relationships between the inputs and outputs through a training phase that mimics the way of human brain activities [41, 65]. For illustration, a fully connected DNN with depth L is depicted in Fig. 2. Therein, the first layer is known as the input layer which consists of three neurons corresponding with the spatial coordinates (x, y, z) . The middle layers are referred to as hidden layers. Finally, the last layer is called the output layer with one neuron which corresponds to the predicted cross-sectional area (\hat{A}) . And meanwhile, the number of hidden layers and hidden neurons depend on the complexity of the application.

Fig. 2 Architecture of a fully connected DNN



Furthermore, the units of the present layer are connected to all units in the previous layer via weights \mathbf{W} and biases \mathbf{b} .

In order to train the network, the feedforward and backpropagation algorithms will be employed to tune the parameters. Specifically, in the feedforward process, a mapping from input to output nodes can be expressed as $\hat{\mathbf{I}}: \mathbb{R}^3 \rightarrow \mathbb{R}$. The data is transmitted from the first layer to the last layer by the transformations. Hence, the relation between the input and output of the k th hidden layer is expressed as

$$\begin{aligned} \hat{\mathbf{I}}^h &= \mathbf{W}^{hT} \hat{\mathbf{o}}^{h-1} + \mathbf{b}^h, \\ \hat{\mathbf{o}}^h &= f(\hat{\mathbf{I}}^h), \end{aligned} \tag{5}$$

where $f(\cdot)$ is the activation function, which allows for learning a complex relationship between input and output. There are several common choices, including ReLU, LeakyReLU, Sigmoid, Softmax, and Tanh. $\hat{\mathbf{o}}^h$ denotes the output of the h th hidden layer. And, the output of each layer can be rewritten as follows

$$\begin{aligned} \text{input layer} &: \hat{\mathbf{o}}^0 = [x, y, z] \in \mathbb{R}^3, \\ \text{hidden layers} &: \hat{\mathbf{o}}^h = f(\mathbf{W}^{hT} \hat{\mathbf{o}}^{h-1} + \mathbf{b}^h) \in \mathbb{R}^{m_h}, \\ &\text{for } 1 \leq h \leq (L - 1), \\ \text{output layer} &: \hat{\delta}^L = f(\mathbf{W}^{LT} \hat{\mathbf{o}}^{L-1} + \mathbf{b}^L) = \hat{A} \in \mathbb{R}, \end{aligned} \tag{6}$$

where m_h is the number of neurons in the h th hidden layer.

In order to achieve the optimal parameters, the training is carried out by minimizing the loss function \mathcal{L} . There are many available optimizers to train the DNN model. Adam optimizer is one of the most common ones, so it is utilized to perform the training task. Accordingly, the network parameters at timestep t are updated as follows

$$\boldsymbol{\theta}_t = \boldsymbol{\theta}_{t-1} - \eta \frac{\hat{\mathbf{m}}_t}{\sqrt{\hat{\mathbf{v}}_t + \epsilon}}, \tag{7}$$

where $\boldsymbol{\theta}$ is the parameter vector including weights and biases of the network; η is the learning rate; ϵ is a constant added to maintain numerical stability; $\hat{\mathbf{m}}_t$ and $\hat{\mathbf{v}}_t$ are the bias-corrected first and second raw moment, and they are given by

$$\hat{\mathbf{m}}_t = \frac{\mathbf{m}_t}{1 - \beta_1^t}, \tag{8}$$

$$\hat{\mathbf{v}}_t = \frac{\mathbf{v}_t}{1 - \beta_2^t}, \tag{9}$$

in which $\beta_1, \beta_2 \in [0, 1)$ are the hyper-parameters to control \mathbf{m}_t and \mathbf{v}_t which are two exponential decay rates at timestep t as follows

$$\mathbf{m}_t = \beta_1 \mathbf{m}_{t-1} + (1 - \beta_1) \cdot \nabla \mathcal{L}(\boldsymbol{\theta}_{t-1}), \tag{10}$$

$$\mathbf{v}_t = \beta_2 \mathbf{v}_{t-1} + (1 - \beta_2) \cdot \nabla \mathcal{L}(\boldsymbol{\theta}_{t-1}), \tag{11}$$

Here, $\nabla \mathcal{L}(\boldsymbol{\theta}_{t-1})$ is the first gradient of the loss function with respect to the parameters at timestep $(t - 1)$. In this work, Adam optimizer with its default parameters as suggested by Kingma and Ba [66] was used to train the model. For more details, interested readers are suggested to refer to Ref. [66].

3.3 Loss function

It should be noted that the proposed approach is designed based on unsupervised learning. Consequently, the predicted cross-sectional area value $\hat{A}_i(\boldsymbol{\theta})$ is expressed by the parameters of the DNN. The loss function is defined as the penalty function in Eq. (2). It includes the mass and responses of structure obtained by FEA corresponding to the predicted cross-sectional areas $\hat{\mathbf{A}}(\boldsymbol{\theta})$, as expressed below

$$\mathcal{L}(\boldsymbol{\theta}) = (1 + \epsilon_1 c)^{\epsilon_2} W(\hat{\mathbf{A}}(\boldsymbol{\theta})). \tag{12}$$

It is easily seen that instead of solving the optimization problem in Eq. (1), we now turn to minimize the loss function by training to find the optimal parameters $\boldsymbol{\theta}^*$ of the network.

$$\boldsymbol{\theta}^* = \arg \min_{\boldsymbol{\theta}} (\mathcal{L}(\boldsymbol{\theta})). \tag{13}$$

Once the network is trained, the optimum cross-sectional areas of truss members are found corresponding to the optimal parameters. Note that the derivatives of the loss function with respect to the parameters θ_i obtained by applying the chain rule to Eq. (12) is given by

$$\frac{\partial \mathcal{L}}{\partial \theta_i} = \sum_{e=1}^m \frac{\partial \mathcal{L}}{\partial \hat{A}_e} \frac{\partial \hat{A}_e}{\partial \theta_i}. \tag{14}$$

From Eq. (14), it can be observed that the term $\frac{\partial \hat{A}_e}{\partial \theta_i}$ will be calculated automatically by the backpropagation algorithm, and the other term $\frac{\partial \mathcal{L}}{\partial \hat{A}_e}$ must be provided as follows

$$\frac{\partial \mathcal{L}}{\partial \hat{A}_e} = \epsilon_2 (1 + \epsilon_1 c)^{\epsilon_2 - 1} W \frac{\partial c}{\partial \hat{A}_e} + (1 + \epsilon_1 c)^{\epsilon_2} \frac{\partial W}{\partial \hat{A}_e}. \tag{15}$$

Here, the derivative of the sum of the violated constraints $\frac{\partial c}{\partial \hat{A}_e}$ is defined as

$$\frac{\partial c}{\partial \hat{A}_e} = \sum_{j=1}^n \frac{\partial}{\partial \hat{A}_e} \max(0, g_j), \tag{16}$$

with

$$\frac{\partial}{\partial \hat{A}_e} \max(0, g_j) = \begin{cases} 0 & g_j \leq 0, \\ \frac{\partial g_j}{\partial \hat{A}_e} & g_j > 0, \end{cases} \tag{17}$$

in which g_j is the j th constraint function. In this work, three types of constraints are considered including displacement, stress, and frequency. As pointed out by Camarda [67] and Iranmanesh [42], the gradient of these constraints with respect to the area of the members is given by

$$\frac{\partial \mathbf{u}}{\partial A_e} = -\mathbf{K}^{-1} \frac{\partial \mathbf{K}}{\partial A_e} \mathbf{u}, \quad (18)$$

$$\frac{\partial \sigma_e}{\partial \hat{A}_e} = \frac{E}{L_e} \mathbf{T} \frac{\partial \mathbf{u}_e}{\partial \hat{A}_e}, \quad (19)$$

$$\frac{\partial \omega_f}{\partial \hat{A}_e} = \boldsymbol{\phi}_f^T \frac{\partial \mathbf{K}}{\partial \hat{A}_e} \boldsymbol{\phi}_f - \omega_f^2 \boldsymbol{\phi}_f^T \frac{\partial \mathbf{M}}{\partial \hat{A}_e} \boldsymbol{\phi}_f, \quad (20)$$

where \mathbf{u} is the displacement vector; \mathbf{K} is the global stiffness matrix; σ_e stands for the stress of the member; \mathbf{T} is the geometric transformation matrix; \mathbf{u}_e denotes the displacement vector of the element; ω_f and $\boldsymbol{\phi}_f$ are the angular frequency and the eigenvalue vector, respectively; \mathbf{M} stands for the mass matrix. As can be seen, the method for computing the sensitivity of constraints can be directly determined from

Eqs. (16)–(20), but it takes effort to achieve success. To circumvent this, the automatic differentiation tool JAX is utilized in this work for accelerated performance [59]. It has been successfully applied to the topology optimization problem [68]. In contrast, the gradient of the objective function is easily defined as follows

$$\frac{\partial W}{\partial \hat{A}_e} = \rho_e \sum_{k=1}^{ng} L_k, \quad (21)$$

where W is the structural weight; ng denotes the total number of member groups, and ρ_e stands for the density of members.

According to the afore-presented issue, the gradient of the loss function is easily estimated. Hence, the parameters of the network are tuned for each epoch of the training process. The main steps of the proposed approach are summarized in Algorithm 1. It is worth mentioning that this approach is based on the gradient descent algorithm and the optimization problem is a nonconvex one [2, 3]. Therefore, we need to pay attention to the sensitivity of the initial parameters as well as to the possibility of being stuck in local minima and saddle points.

Algorithm 1: Optimization of truss structures using deep unsupervised learning-based framework

Input:

- Structure: geometry, material properties, loads, boundary conditions
- Network: number of hidden layers, hidden neurons, activation function, Adam optimizer

Output: optimal parameters $\boldsymbol{\theta}$, optimum weight of truss structures

- 1 Build a DNN with uniformly distributed initial parameters $\boldsymbol{\theta}_0$ from $[-1, 1]$
 - 2 Set the default values of Adam optimizer parameters [66]
 - 3 **while** $\|\nabla \mathcal{L}(\boldsymbol{\theta})\| > 0.01$ or *Maxepoch* is not reached **do**
 - 4 $t = t + 1$
 - 5 Predict $\hat{\mathbf{A}}$ using the feedforward phase
 - 6 Compute the weight $W(\hat{\mathbf{A}})$ of truss structure
 - 7 Evaluate the constraint function $g_j(\hat{\mathbf{A}})$ by FEA
 - 8 Loss function $\mathcal{L}(\boldsymbol{\theta}_{t-1})$ is estimated by Eq. (12)
 - 9 $\frac{\partial W}{\partial \hat{A}_e}$ is calculated by Eq. (21)
 - 10 $\frac{\partial \sigma_e}{\partial \hat{A}_e}$ is determined by the automatic differentiation JAX
 - 11 $\frac{\partial \omega_f}{\partial \hat{A}_e}$ is calculated automatically by the backpropagation algorithm
 - 12 Calculate the sensitivity of loss function $\frac{\partial \mathcal{L}}{\partial \theta_i}$ by Eq. (14)
 - 13 Update parameters $\boldsymbol{\theta}_t$ of the network by Eq. (7)
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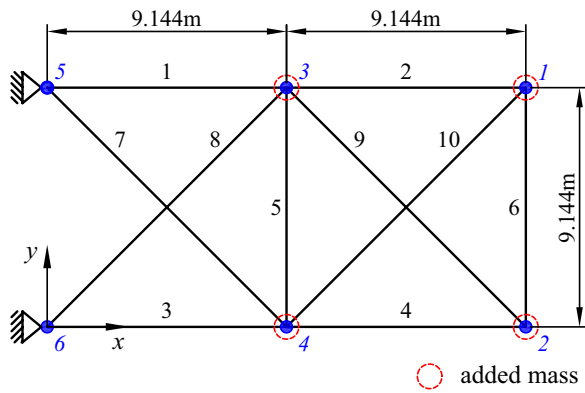


Fig. 3 10-bar planar truss problem

4 Numerical experiments

In this section, several well-known benchmark problems are explored to justify the efficiency of the proposed approach, and obtained results are compared with the DE and other metaheuristic algorithms done by previous works. Depending on the constraints’ characteristics, the examples are divided into two categories: the first one is made up of four cases of natural frequency constraints, while the remaining one includes other three problems with displacement and stress constraints. In all investigated examples, two and three neurons are used in the input layer corresponding to 2D and 3D problems, and one neuron with regard to members’ cross-sectional area is utilized for the output layer. Besides,

the trial-error and grid search (GS) methods [69] are applied to estimate the architecture of the network. In addition, the Adam optimizer (learning rate 0.01, $\beta_1=0.9$, $\beta_2=0.999$) [66] is adopted for training the model. LeakyReLU and Softmax are activation functions for the hidden and output layer, respectively. Otherwise, the learning process of the network is terminated when either the norm of the gradient value is less than 0.01 or the maximum number of epoch reaches [70]. Note that for the present method, the number of epochs is also the number of FEAs or loss function evaluations.

For DE algorithm, the parameters are set as follows: population size 20, the maximum number of FEAs 3000, mutant factor $F = 0.8$, crossover control parameter $Cr = 0.9$, and the value of threshold 10^{-6} [2, 10, 60]. The best design is determined by 30 independent runs of each problem, dealing with the stochastic nature of the DE algorithm. To get a fair comparison between the different methods, all the tests are implemented in the library of Tensorflow and using Python software. And all experiments are performed on a desktop computer with Core i5-8500 CPU of @3.0 GHz and 16 GB RAM.

4.1 Frequency constraints

4.1.1 10-bar planar truss

The first numerical example is a 10-bar planar truss structure, as illustrated in Fig. 3. The data for the design, including Young’s modulus, density, design variable bounds, and natural frequency constraints, are summarized in Table 1.

Table 1 Material properties, constraints, upper and lower bounds on area variables

Test problems	Young’s modulus E (N/m ²)	Material density ρ (kg/m ³)	Natural frequency (Hz)	Cross-sectional area limits (cm ²)
10-bar planar truss	6.98×10^{10}	2770	$7 \leq f_1; 15 \leq f_2; 20 \leq f_3$	$0.645 \leq A_i \leq 50$
72-bar space truss	6.98×10^{10}	2770	$f_1 = 4; 6 \leq f_3$	$0.645 \leq A_i \leq 20$
120-bar dome truss	2.1×10^{11}	7971.81	$9 \leq f_1; 11 \leq f_2$	$1 \leq A_i \leq 129.3$
200-bar planar truss	2.1×10^{11}	7860	$5 \leq f_1; 10 \leq f_2; 15 \leq f_3$	$0.1 \leq A_i \leq 25$

Table 2 MSE of the optimal cross-sectional areas for the 10-bar planar truss with various optimizers and activation functions

Activation functions	Optimizers							
	Adam	Adamw	Adadelta	Adagrad	Adamax	SGD	ASGD	RMSprop
Softmax	0.0466	0.0610	129.7283	20.4489	6.2696	87.8649	204.8981	7.3892
Softplus	5.7412	5.4541	45.3525	6.0972	5.7074	6.2923	31.0242	22.5772
Tanh	0.0918	0.0529	30.8862	7.7082	0.1728	9.0307	19.7944	7.7507
Sigmoid	6.5302	6.4860	111.7375	6.8888	5.8379	22.6053	99.6836	14.6526
ReLU	0.1864	7.2302	33.8680	6.6109	6.6240	6.2492	19.6532	13.2780
LeakyReLU	0.0247	0.0332	33.3514	6.3805	7.4386	6.3581	20.6535	11.5345

Table 3 Standard errors (%) of weight and frequencies for the 10-bar planar truss with various activation functions using Adam

Error (%)	Activation functions					
	Softmax	Softplus	Tanh	Sigmoid	ReLU	LeakyReLU
Weight	0.3489	1.7766	0.3596	1.5482	0.2001	0.0487
f_1	0.0281	0.0028	0.0432	0.0398	0.0196	0.0014
f_2	2.6639	11.2745	3.1186	11.2354	0.8680	0.1062
f_3	0.1181	0.1950	0.0029	0.0995	0.0949	0.0380

Table 4 Optimization results obtained for the 10-bar planar truss with frequency constraints

Design variables	PSO	FA	DPSO	OC-GA	IDE	AHEFA	This study	
	[11]	[9]	[12]	[14]	[2]	[10]	DE	DUL
A_i (cm ²)								
A_1	37.712	36.198	35.944	37.284	35.060	35.171	35.288	35.106
A_2	9.959	14.030	15.530	9.445	14.685	14.720	14.577	14.826
A_3	40.265	34.754	35.285	35.051	35.068	35.107	35.087	35.227
A_4	16.788	14.900	15.385	19.262	14.809	14.698	14.782	14.723
A_5	11.576	0.654	0.648	2.783	0.645	0.645	0.645	0.654
A_6	3.955	4.672	4.583	5.450	4.558	4.559	4.561	4.568
A_7	25.308	23.467	23.610	19.041	23.527	23.733	23.510	23.712
A_8	21.613	25.508	23.599	27.939	23.799	23.679	23.972	23.712
A_9	11.576	12.707	13.135	14.950	12.503	12.398	12.462	12.371
A_{10}	11.186	12.351	12.357	10.361	12.459	12.423	12.266	12.371
Best weight (kg)	537.980	531.280	532.390	535.730	524.462	524.451	524.463	524.719
Weight error (%)	2.580	1.302	1.514	2.151	0.002	–	0.002	0.051
f_1 (Hz)	7.000	7.000	7.000	7.001	7.000	7.000	7.000	7.000
f_2 (Hz)	17.786	16.164	16.187	17.030	16.185	16.192	16.196	16.213
f_3 (Hz)	20.000	20.003	20.000	20.156	20.000	20.000	20.000	20.008
Number of FEAs	–	5000	3000	12,000	6260	5860	9480	1000
Total times (s)	–	–	–	–	–	–	72.134	29.784

Bold used to emphasize the best minimum weight design

A lumped mass of 454 kg is set at four free nodes. This benchmark problem has been previously analyzed by many researchers using metaheuristic algorithms, such as Gomes [11], Kaveh [12], Ho [2], and Lieu [10], etc. For the first case, a DNN with five hidden layers is chosen to build the model and each hidden layer contains 20 neurons. The model performs training with the maximum number of FEAs equal to 1,000.

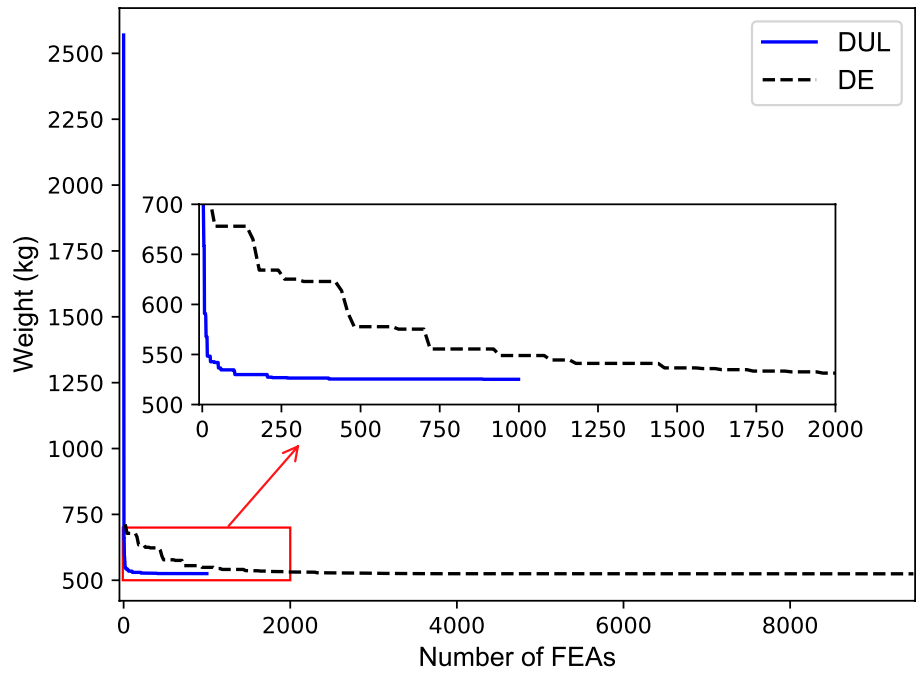
With the above architecture, a survey is conducted to investigate the effect of different combinations of optimizers and activation functions on the performance of the network. Herein, mean square error (MSE) of the optimal solution between the DUL and DE is employed as the standard tool to measure. Table 2 summarizes the obtained MSE results. It can be seen that Adam and Adadelta are the best and worst optimizers, respectively. Therein, the combination of Adam and LeakyReLU provides the lowest MSE (0.0247). In addition, Table 3 shows the errors of the objective and constraints compared with the DE when Adam is combined with various activation functions. It is clear that LeakyReLU is the lowest error value and less than 0.2%. Consequently,

Adam and LeakyReLU are respectively chosen as optimizer and activation function for the training network in this study.

Table 4 shows a comparison of optimal results gained by this work and other researchers. It can be observed that the optimum weight acquired by DUL (524.719 kg) is very close to DE (524.463 kg), AHEFA (524.451 kg), and IDE (524.462 kg). However, the DUL outperforms other well-known existing algorithms available in the literature. More specifically, it requires the least number of structural analyses with only 1000 analyses, while the metaheuristic algorithms require a larger number of FEA calls, e.g. AHEFA [10] with 5860 analyses, DE with 9480 analyses. This can easily be explained by the fact that the training algorithm is used as a gradient-based optimization of stochastic loss function, so the number of function evaluations will be drastically reduced. All frequency constraints are satisfied with the small relative errors. In addition, the weight error against to the AHEFA is small (0.051%), which is only higher than DE (0.002%) and IDE (0.002%).

The weight convergence histories of the DUL and DE are depicted in Fig. 4. As observed, the proposed DUL model's

Fig. 4 The weight convergence histories obtained using the DUL and DE for the 10-bar planar truss



learning curve shows that the weight of structure rapidly decreases at the beginning, tends to be stable around 400 analyses, and reaches the optimal solution only through 1000 analyses. On the contrary, the traditional DE converges much slower and requires more than 9 times the number of analyses (9480 analyses) compared to our model. In addition, this work only requires one run time with 29.784 s, while DE approximately takes 72 s to complete 30 independent optimization runs to get as accurate results as possible. Therefore, our approach helps to save more than a half of the computational cost.

4.1.2 72-bar space truss

Next, a 72-bar space truss structure, as shown in Fig. 5, is investigated as the second example for size optimization. Members’ cross-sectional areas are classified into 16 categories, which corresponds to the 16 design variables as listed in Table 5. A lumped mass of 2270 kg is mounted at each of the four upper nodes of the structure. The data concerning the design of this example is contained in Table 1. This structure has been formally investigated by various researchers [2, 9–11, 60]. A 6-layer network (3-20-20-20-20-1) is trained with maximum epochs of 3000.

A comparison of the optimal results attained by the proposed approach and previous works is illustrated in Table 5. It can be observed that the optimal weight (325.161 kg) found by the DUL is close to AHEFA (324.237 kg) with the error less than 0.3%, and smaller than the other studies (PSO [11]: 328.823 kg; FA [9]: 328.334 kg; and HALC-PSO [60]:

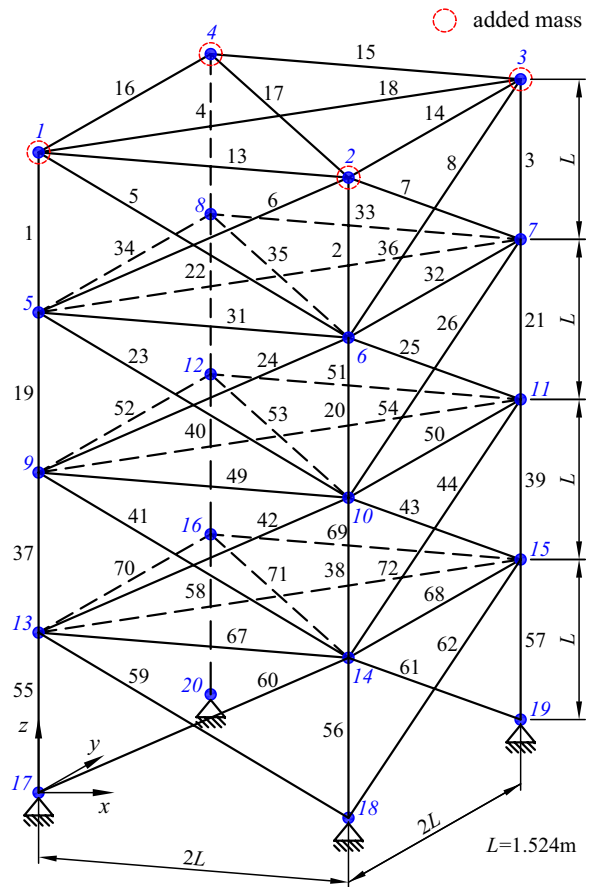


Fig. 5 Schematic of a 72-bar space truss structure

Table 5 Comparison of the obtained results for the 72-bar space truss with frequency constraints

Design variables	PSO	FA	HALC-PSO	IDE	AHEFA	This study	
						DE	DUL
A_i (cm ²)	[11]	[9]	[60]	[2]	[10]		
A_1 – A_4	2.987	3.6803	3.3437	3.5863	3.5612	3.6191	3.8710
A_5 – A_{12}	7.849	7.6808	7.8688	7.8278	7.8736	7.8203	7.8864
A_{13} – A_{16}	0.645	0.6450	0.6450	0.6450	0.6450	0.6453	0.6495
A_{17} – A_{18}	0.645	0.6450	0.6450	0.6450	0.6451	0.6452	0.6456
A_{19} – A_{22}	8.765	9.4955	8.1626	8.1052	7.9710	7.9086	8.4929
A_{23} – A_{30}	8.153	8.2870	7.9502	7.8788	7.8928	7.9904	8.0346
A_{31} – A_{34}	0.645	0.6450	0.6452	0.6451	0.6450	0.6451	0.6522
A_{35} – A_{36}	0.645	0.6461	0.6452	0.6450	0.6451	0.6466	0.6482
A_{37} – A_{40}	13.450	11.4510	12.2668	12.5157	12.5404	12.6016	12.9743
A_{41} – A_{48}	8.073	7.8990	8.1845	8.0102	7.9639	7.8204	8.0028
A_{49} – A_{52}	0.645	0.6473	0.6451	0.6450	0.6459	0.6451	0.6462
A_{53} – A_{54}	0.645	0.6450	0.6451	0.6452	0.6462	0.6455	0.6450
A_{55} – A_{58}	16.680	17.4060	17.9632	16.9997	17.1323	17.0752	15.9486
A_{59} – A_{66}	8.159	8.2736	8.1292	8.0362	8.0216	8.1264	7.9057
A_{67} – A_{70}	0.645	0.6450	0.6450	0.6451	0.6450	0.6463	0.6901
A_{71} – A_{72}	0.645	0.6450	0.6450	0.6453	0.6451	0.6461	0.6455
Best weight (kg)	328.823	328.334	327.770	324.244	324.237	324.289	325.161
Weight error (%)	1.414	1.264	1.090	0.002	–	0.016	0.285
f_1 (Hz)	4.00	4.00	4.00	4.00	4.00	4.00	4.00
f_3 (Hz)	6.00	6.00	6.00	6.00	6.00	6.00	6.01
Number of FEAs	–	50,000	8000	11,620	8860	11,740	3000
Total times (s)	–	–	–	–	–	236.763	148.489

Bold used to emphasize the best minimum weight design

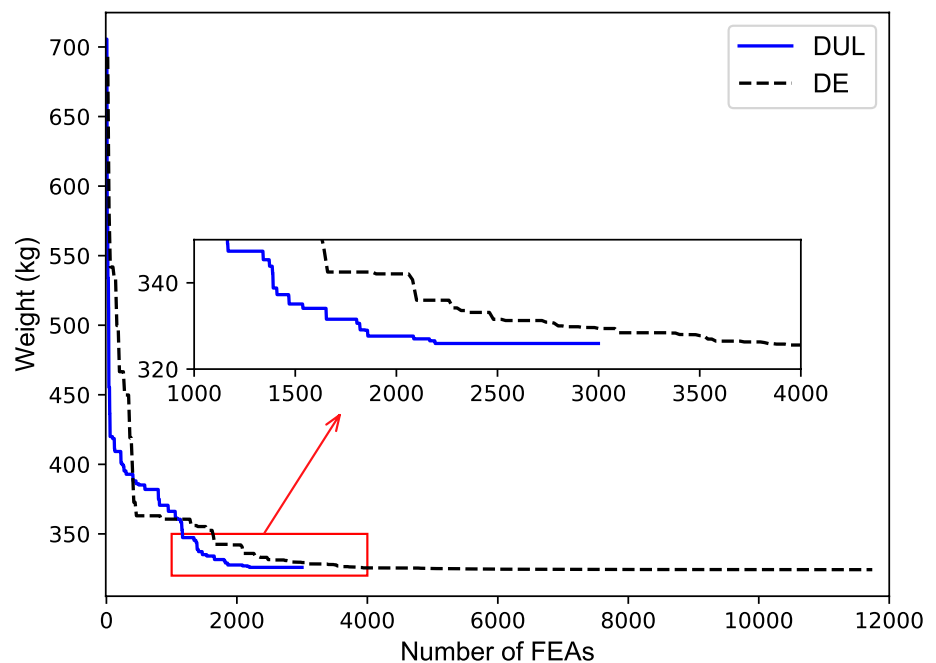
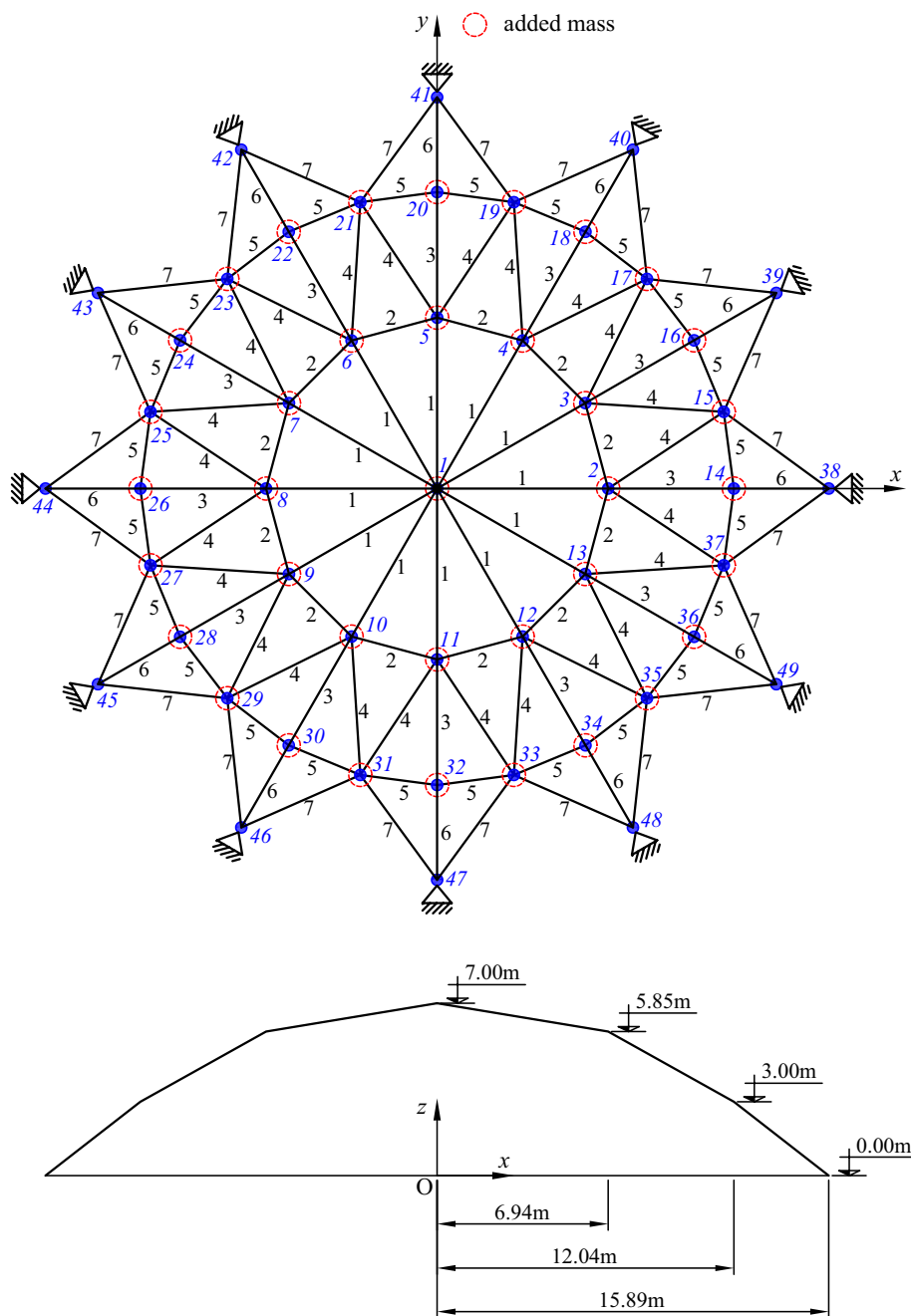
Fig. 6 The weight convergence histories obtained using the DUL and DE for the 72-bar space truss

Fig. 7 A 120-bar dome truss structure



327.77 kg). Compared to the computational cost, the DUL required less structural analyses than other optimization techniques, namely 74% DE, 66% AHEFA [10], 74% IDE [2], 62% HALC-PSO [60], and 94% FA [9], but the frequencies still satisfy constraints. Furthermore, it takes 148.489 s for one run to achieve the near-global optimal solution, while DE needs 236.763 s with 30 runs to get the goal. Again, DUL shows its efficiency in significantly reducing the computational effort. Figure 6 shows the weight convergence histories obtained using two different approaches. Clearly, the convergence rate of the DUL is much faster than that of

the DE algorithm. The weight obtained by DUL approaches the optimal solution after only 3,000 analyses, while the DE is still far behind (11,740 analyses).

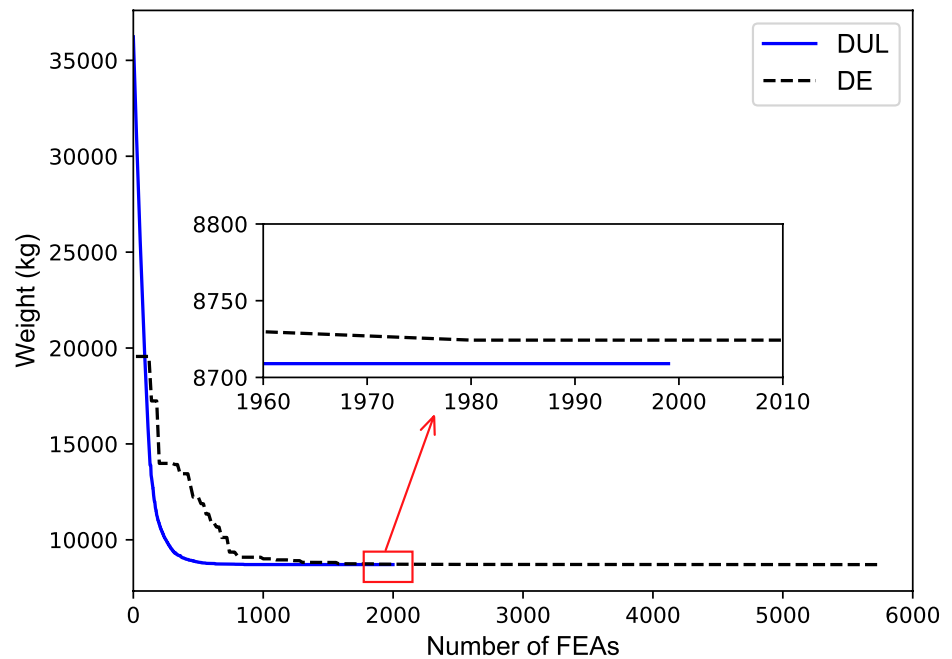
4.1.3 120-bar dome truss

A dome truss with 120 members illustrated in Fig. 7 is examined. Here, $m_1 = 3000$ kg, $m_2 = 500$ kg, and $m_3 = 100$ kg are the lump masses which are attached to node 1, nodes 2–13, and the remaining nodes, respectively. Design variable bounds, material properties, and constraints for this example

Table 6 Optimization results obtained for the 120-bar dome truss with frequency constraints

Design variables	DPSO	OMGSA	HALC-PSO	IDE	AHEFA	This study	
						DE	DUL
A_i (cm ²)	[12]	[71]	[60]	[2]	[10]		
A_1	19.607	20.263	19.8905	19.4670	19.5094	19.4359	19.5002
A_2	41.290	39.294	40.4045	40.5004	40.3867	40.5901	40.4118
A_3	11.136	9.989	11.2057	10.6136	10.6033	10.5945	10.6020
A_4	21.025	20.563	21.3768	21.1073	21.1168	21.0916	21.1160
A_5	10.060	9.603	9.8669	9.8417	9.8221	9.8641	9.8428
A_6	12.758	11.738	12.7200	11.7735	11.7735	11.8052	11.7916
A_7	15.414	15.877	15.2236	14.8264	14.8405	14.8191	14.8365
Best weight (kg)	8890.480	8724.970	8889.960	8707.289	8707.255	8707.404	8708.975
Weight error (%)	2.104	0.203	2.098	0.000	–	0.002	0.020
f_1 (Hz)	9.0001	9.0020	9.0000	9.0000	9.0000	9.0000	9.0008
f_2 (Hz)	11.0007	11.0030	11.0000	11.0000	11.0000	10.9999	11.0004
Number of FEAs	6000	242,700	17,000	4060	3560	5740	2000
Total times (s)	–	–	–	–	–	1366.651	215.346

Bold used to emphasize the best minimum weight design

Fig. 8 The weight convergence histories obtained using the DUL and DE for the 120-bar dome truss

are tabulated in Table 1. The DNN model contains 2 hidden layers, 20 neurons in each hidden layer and 2000 epochs is the finest in its performance for this particular application.

Table 6 provides a comparison between this study and other studies available in the literature. It is easily seen that the optimum weight obtained by the DUL (8708.975 kg) is smaller than DPSO (8890.48 kg), OMGSA (8724.97 kg), and HALC-PSO (8889.96 kg). Although the obtained results by the DE (8707.404 kg), IDE [2] (8707.289 kg), and AHEFA [10] (8707.255 kg) are slightly lighter than the present method, the errors between them and AHEFA are less than 0.03%. However, it is clear that DUL outperforms

the existing algorithms in terms of computational cost. Our model rapidly finds the optimum weight with only 2000 FEAs, whilst the others demand a larger number of FEAs for the convergence process, i.e. OMGSA with 242,700 analyses, IDE with 4060 analyses. In addition, the computation time of the proposed method (215.346 s) is 6 times less than that of the DE (1366.651 s). Figure 8 depicts the weight convergence histories obtained for this structure using the DUL and DE. Again, the DUL shows its efficiency in significantly improving the convergence rate compared with DE but still ensuring the quality of solution.

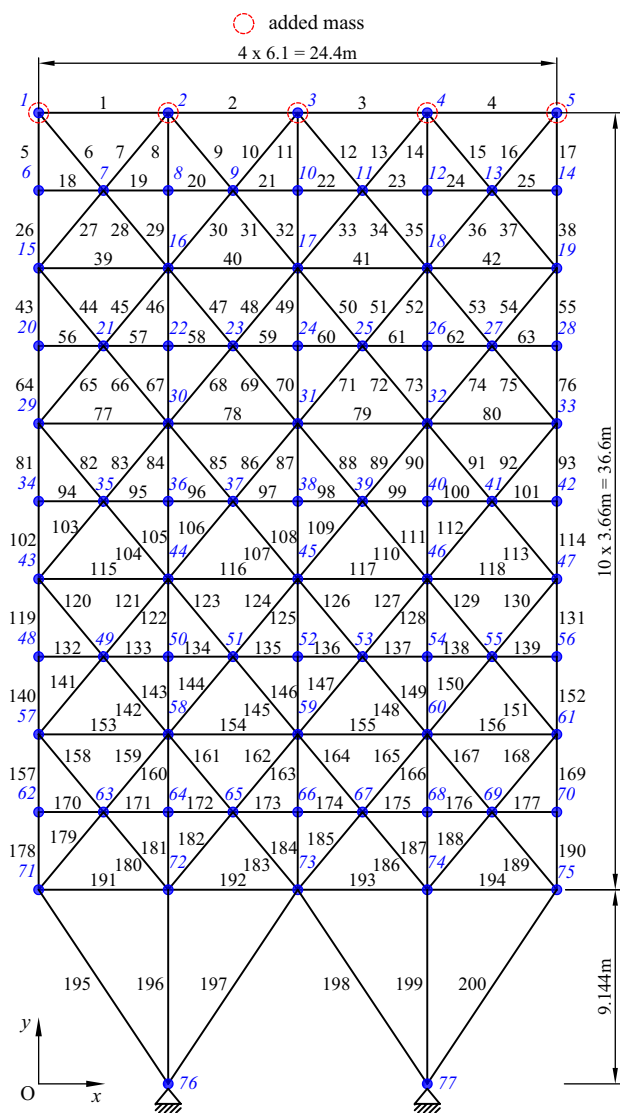


Fig. 9 A 200-bars planar truss structure

4.1.4 200-bar planar truss

The next example deals with the design of a 200-bars planar truss structure as shown in Fig. 9. All members are categorized into 29 groups using symmetry. The grouping information is indicated in Table 7. A lumped mass of 100 kg is added to 5 upper nodes. The data of the optimization is given in Table 1. The DNN uses 5 hidden layers with 20 neurons per layer and a maximum epoch of 2500 for training.

The optimal results, including the design variables, weight as well as frequencies, are tabulated as Table 8. As expected, the obtained results of the proposed model (2161.9 kg) reveals a quite good agreement with the DE (2160.94 kg) and AHEFA [10] (2160.74 kg). Note that although the optimum weight provided by the HALC-PSO [60] (2156.73 kg) and OMGSA [71] (2158.64 kg) are

smaller than those by the DUL, the number of structural analyses is larger than those of the proposed approach. More concretely, the DUL only requires 2500 analyses for the convergence performance, while the DE, AHEFA, OMGSA [71], PFJA [72], and HALC-PSO [60] take 20,000, 11,300, 7252, 10,546, and 13,000, respectively. Additionally, DUL (513.965 s) only spends one-tenth of the computation time of DE (6375.934) to obtain the near-global optimal solution with the error less than 0.3%. Along with the results of the above mentioned examples, the proposed approach shows a significant saving of computational efforts when the complexity of problems is increased. The convergence curves in Fig. 10 gives a more detailed performance view for the present work. Clearly, this procedure converges very quickly to the near-optimal solution, while the DE is still quite far from the target value.

4.2 Stress and displacement constraints

4.2.1 25-bar space truss

A 25-bars space truss shown in Fig. 11 is investigated for the design optimization with displacement and stress constraints. The material density and Young’s modulus are 0.1 lb/in³ and 10⁴ ksi. It is subjected to concentrated loads $P_y = 20$ kips and $P_z = 5$ kips as shown in Fig. 11. Cross-sectional areas of all members are divided into 8 groups corresponding to 8 design variables with the minimum gauge of 0.01 in² and the allowable stress of groups are listed in Table. 9. All displacements of nodes are limited in interval $[-0.35, 0.35]$ in. In addition, the network architecture including 3 hidden layers, 20 neurons in each hidden layer, and 1500 epochs is used to build the network.

As the previously presented examples, the optimal results gained by this study in comparison with other studies are summarized in Table 10. It is easily seen that the optimum weight found by the DUL (545.7 kg) agrees well with the DE (545.16 kg), TLBO [75] (545.09 kg), HPSO [74] (545.19 kg), and BB-BC [73] (545.38 kg) without violating the design constraints, while the lighter designs obtained by the HS [16] (544.38 kg) and HBB-BC [3] (545.16 kg) violated the design constraints. Furthermore, the DUL only requires 1500 structural analyses with the small error 0.113%, and much less than HS [16] (15,000), BB-BC [73] (20,566), HPSO [74] (125,000), HBB-BC [3] (12,500), TLBO [75] (15,318) and DE (8480). In addition, the total times of DUL (49.470 s) is less than a half of DE (104.069 s). Obviously, the current approach requires the lower computational cost compared to the conventional algorithms. Figure 12 illustrates the weight convergence histories using the DUL and DE. One again shows the feature of the algorithm that the convergence speed accelerates in the early stage and quickly stabilizes after only 1200 analyses.

Table 7 Design variables of the 200-bar planar truss

Design variables	Member group	Design variables	Member group
A ₁	1, 2, 3, 4	A ₁₆	82, 83, 85, 86, 88, 89, 91, 92, 103, 104, 106, 107, 109, 110, 112, 113
A ₂	5, 8, 11, 14, 17	A ₁₇	115, 116, 117, 118
A ₃	19, 20, 21, 23, 24	A ₁₈	119, 122, 125, 128, 131
A ₄	18, 25, 56, 63, 94, 101, 132, 139, 170, 177	A ₁₉	133, 134, 135, 136, 137, 138
A ₅	26, 29, 32, 35, 38	A ₂₀	140, 143, 146, 149, 152
A ₆	6, 7, 9, 10, 12, 13, 15, 16, 27, 28, 30, 31, 33, 34, 36, 37	A ₂₁	120, 121, 123, 124, 126, 127, 129, 130, 141, 142, 144, 145, 147, 148, 150, 151
A ₇	39, 40, 41, 42	A ₂₂	153, 154, 155, 156
A ₈	43, 46, 49, 52, 55	A ₂₃	157, 160, 163, 166, 169
A ₉	57, 58, 59, 60, 61, 62	A ₂₄	171, 172, 173, 174, 175, 176
A ₁₀	64, 67, 70, 73, 76	A ₂₅	178, 181, 184, 187, 190
A ₁₁	44, 45, 47, 48, 50, 51, 53, 54, 65, 66, 68, 69, 71, 72, 74, 75	A ₂₆	158, 159, 161, 162, 164, 165, 167, 168, 179, 180, 182, 183, 185, 186, 188, 189
A ₁₂	77, 78, 79, 80	A ₂₇	191, 192, 193, 194
A ₁₃	81, 84, 87, 90, 93	A ₂₈	195, 197, 198, 200
A ₁₄	95, 96, 97, 98, 99, 100	A ₂₉	196, 199
A ₁₅	102, 105, 108, 111, 114		

Table 8 Comparison of the obtained results for the 200-bar planar truss with frequency constraints

Design variables	HALC-PSO	PFJA	OMGSA	AHEFA	This study	
					DE	DUL
A_i (cm ²)	[60]	[72]	[71]	[10]		
1	0.3072	0.30785	0.289	0.2993	0.3096	0.3040
2	0.4545	0.47168	0.486	0.4508	0.4485	0.4588
3	0.1000	0.10020	0.100	0.1001	0.1001	0.1016
4	0.1000	0.10010	0.100	0.1000	0.1000	0.1106
5	0.5080	0.54175	0.499	0.5123	0.5114	0.5317
6	0.8276	0.81840	0.804	0.8205	0.8188	0.8247
7	0.1023	0.10096	0.103	0.1011	0.1007	0.1026
8	1.4357	1.42367	1.377	1.4156	1.4241	1.4248
9	0.1007	0.10006	0.100	0.1000	0.1001	0.1026
10	1.5528	1.63199	1.554	1.5742	1.5812	1.5895
11	1.1529	1.13730	1.151	1.1597	1.1662	1.1591
12	0.1522	0.10000	0.131	0.1338	0.1404	0.1472
13	2.9564	2.97378	3.028	2.9672	2.9833	2.9952
14	0.1003	0.10072	0.101	0.1000	0.1005	0.1123
15	3.2242	3.32736	3.261	3.2722	3.2274	3.2739
16	1.5839	1.55580	1.612	1.5762	1.5850	1.5866
17	0.2818	0.23602	0.209	0.2562	0.3016	0.2141
18	5.0696	5.20167	5.020	5.0956	5.0229	5.0919
19	0.1033	0.10000	0.133	0.1001	0.1001	0.1155
20	5.4657	5.47313	5.453	5.4546	5.4589	5.4540
21	2.0975	2.09090	2.113	2.0933	2.0996	2.0911
22	0.6598	0.66730	0.723	0.6737	0.7185	0.6511
23	7.6585	7.53409	7.724	7.6498	7.6974	7.4208
24	0.1444	0.10489	0.182	0.1178	0.1127	0.1880
25	8.0520	7.87075	7.971	8.0682	8.1285	7.7226
26	2.7889	2.81229	2.996	2.8025	2.8448	2.7871
27	10.4770	10.70210	10.206	10.5040	10.3676	10.5569
28	21.3257	21.75078	20.699	21.2935	21.1726	21.5607
29	10.5111	10.44841	11.555	10.7410	10.8075	10.5618
Best weight (kg)	2156.73	2171.34	2158.64	2160.74	2160.94	2161.90
Weight error (%)	-	0.6778	0.0886	0.1861	0.1953	0.2400
f_1 (Hz)	5.0000	5.0000	5.0000	5.0000	5.0000	5.0000
f_2 (Hz)	12.2540	12.3074	12.1180	12.1821	12.2668	12.2679
f_3 (Hz)	15.0440	15.0066	15.0290	15.0160	15.0776	15.1197
Number of FEAs	13,000	10,546	7252	11,300	20,000	2500
Total times (s)	-	-	-	-	6375.934	513.965

Bold used to emphasize the best minimum weight design

4.2.2 72-bar space truss

The next example deals with the optimal design of the 72-bars space truss schematized in Fig. 13. The system is subjected to an external loading of 5 kips in the x - and y -axes' positive directions at node 17. Density and Young's modulus of the material are set as 0.1 lb/in³ and 10⁴ ksi. The cross-sectional areas of members as design variables are categorized into 8 groups. All free node displacements are restricted to ± 0.25 . The stress limitations of all members are ± 25 ksi. The minimum design variables are specified

as 0.1 in². The DNN used for this application consists of 5 hidden layers with 20 hidden neurons in each layer and 1000 epochs.

Table 11 summarizes the results obtained by the DUL and other algorithms reported in recent literatures [3, 16, 73–75]. It is noticed that in this example, the DUL achieves the lightest design overall since the lighter design gained by the HPSO violates the design constraints. Furthermore, the number of structural analyses of the present method is dramatically decreased when compared with previous

Fig. 10 The weight convergence histories obtained using the DUL and DE for the 200-bar planar truss

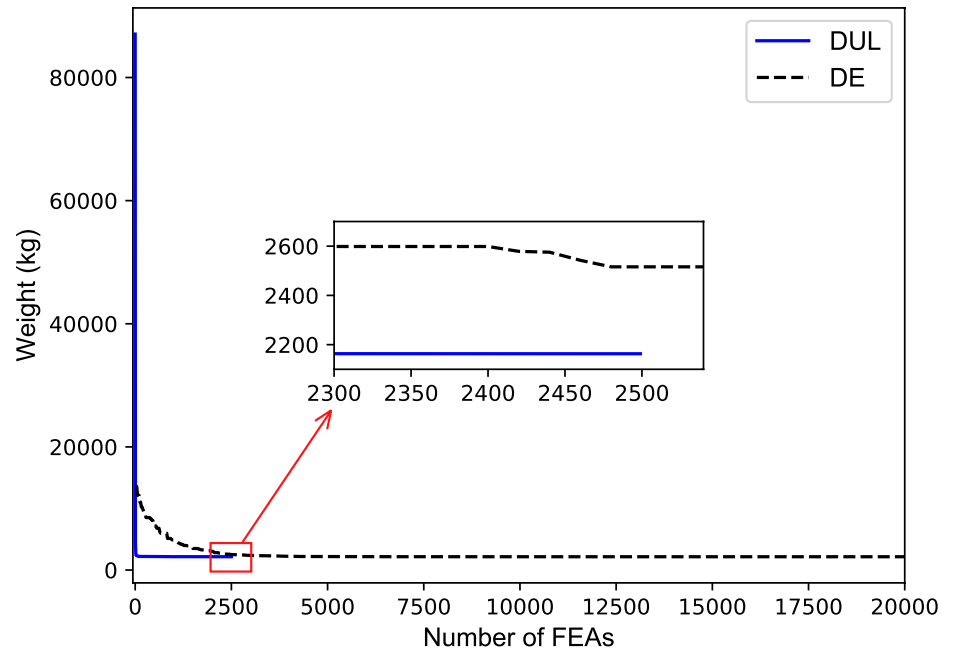


Fig. 11 A 25-bars space truss structure

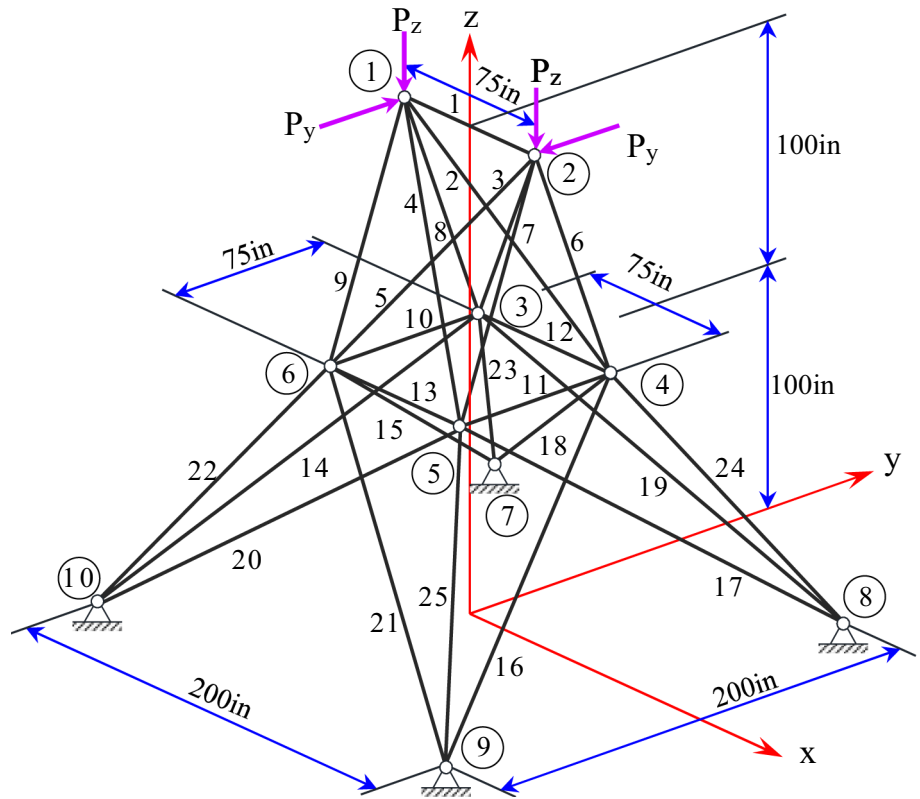


Table 9 Allowable stress values for 25-bar space truss

Design variables	Allowable compressive stress (ksi)	Allowable tension stress (ksi)
A_i (in ²)		
A_1	35.092	40.0
A_2 - A_5	11.59	40.0
A_6 - A_9	17.305	40.0
A_{10} - A_{11}	35.092	40.0
A_{12} - A_{13}	35.092	40.0
A_{14} - A_{17}	6.759	40.0
A_{18} - A_{21}	6.959	40.0
A_{22} - A_{25}	11.082	40.0

studies. More specifically, it takes only 1000 analyses for the convergence performance, while the DE, TLBO [75], HBB-BC [3], HPSO [74], BB-BC [73], and HS [16] take 8580, 19,709, 13,200, 125,000, 19,621, and 20,000, respectively. Moreover, the computational time of DUL (63.715 s) is still much smaller than that of the DE (488.135 s). Finally, Fig. 14 represents the weight convergence histories obtained using two different approaches. As can be seen on the plot, the convergence rate of the learning process is much faster than the traditional DE. Therefore, the present approach has proven again to be the most efficient optimizer.

4.2.3 200-bar planar truss

The last structural optimization problem done herein is to optimize a 200-bar planar truss. The geometry and finite element representation are shown in Fig. 15. As shown in

Table. 7, all cross-section areas of the structure are classified into 29 groups corresponding to design variables. The Young’s modulus is 30,000 ksi and the density is 0.283 lb/in³ for all elements. The allowable stress for all members is set to 10 ksi both in tension and compression. The minimum design variables are 0.1 in². In this example, the displacement constraints are not considered. The structure is designed for three independent loading conditions as following: (1) 1 kip applying in the positive direction of the *x*-axis at nodes 1, 6, 15, 20, 29, 34, 43, 48,57, 62 and 71; (2) 10 kip acting in the negative direction of the *y*-axis at nodes 1, 2, 3, 4, 5, 6, 8, 10, 12, 14, 15, 16, 17, 18, 19, 20,22, 24, 26, 28, 29, 30, 31, 32, 33, 34, 36, 38, 40, 42, 43, 44, 45, 46, 47, 48, 50, 52, 54, 56, 57, 58, 59, 60, 61, 62, 64, 66, 68, 70, 71, 72, 73, 74 and 75; (3) including conditions (1) and (2) acting together. In this case, the network configuration with 5 hidden layers, 20 neurons per layer, and 4,000 epochs are utilized to perform the optimization process.

Table 12 provides a comparison of optimal solutions found by the DUL and several previous studies. It is easily seen that the DUL performs better than the HS [16], HBB-BC [3], CMLPSA [76], and DE in terms of the optimum weight, violated constraints, and needed analyses. The optimum weight obtained by the proposed method is close to the result of the EHS [19], but it requires a smaller number of structural analyses.

It can be observed that the MCOA [15] and TLBO [75] are the first and second-best among the eight algorithms. Note, however, that the selection of control parameter values for the metaheuristic algorithms plays an important role in performance, robustness, and efficiency of them [75, 77–82]. Although, in this case, it was not considered in

Table 10 Optimization results obtained for the 25-bar space truss with displacement and stress constraints

Design variables	HS	BB-BC	HPSO	HBB-BC	TLBO	This study	
						DE	DUL
A_i (in ²)	[16]	[73]	[74]	[3]	[75]		
A_1	0.047	0.010	0.010	2.662	0.0100	0.0100	0.01306
A_2 - A_5	2.022	2.092	1.970	1.993	2.0712	1.9834	1.95147
A_6 - A_9	2.950	2.964	3.016	3.056	2.9570	2.9984	2.96618
A_{10} - A_{11}	0.010	0.010	0.010	0.010	0.0100	0.0100	0.01248
A_{12} - A_{13}	0.014	0.010	0.010	0.010	0.0100	0.0100	0.01282
A_{14} - A_{17}	0.688	0.689	0.694	0.665	0.6891	0.6864	0.69678
A_{18} - A_{21}	1.657	1.601	1.681	1.642	1.6209	1.6776	1.72595
A_{22} - A_{25}	2.663	2.686	2.643	2.679	2.6768	2.6576	2.64302
Best weight (lb)	544.38	545.38	545.19	545.16	545.09	545.16	545.70
Weight error (%)	0.130	0.053	0.018	0.013	–	0.014	0.113
Constraint tolerance (%)	0.206	None	None	2.06	None	None	None
Number of FEAs	15,000	20,566	125,000	12,500	15,318	8480	1500
Total times (s)	–	–	–	–	–	104.069	49.470

Bold used to emphasize the best minimum weight design

Fig. 12 The weight convergence histories obtained using the DUL and DE for the 25-bar planar truss

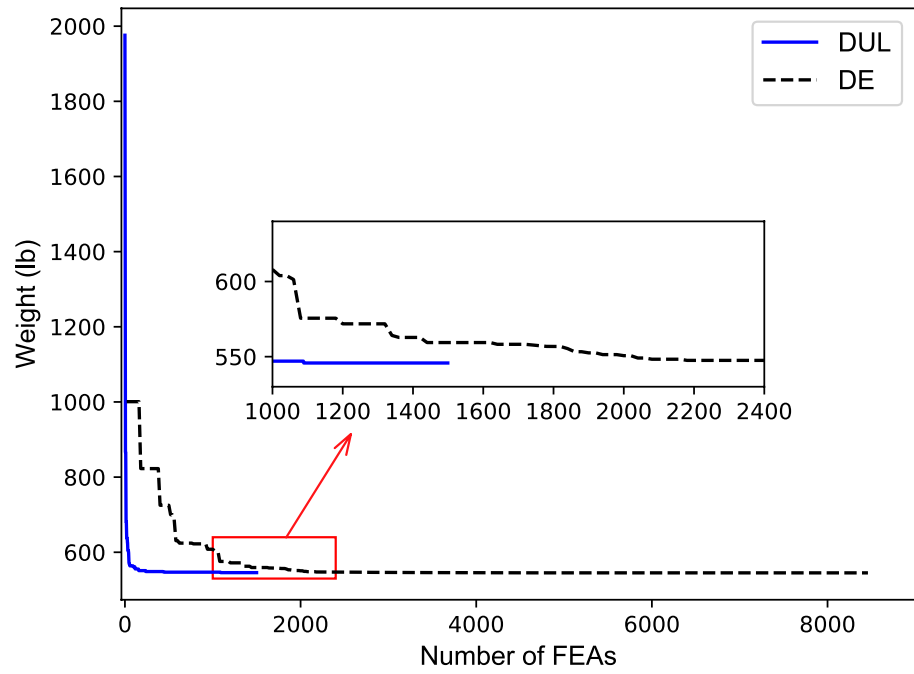


Fig. 13 72-bar space truss structure

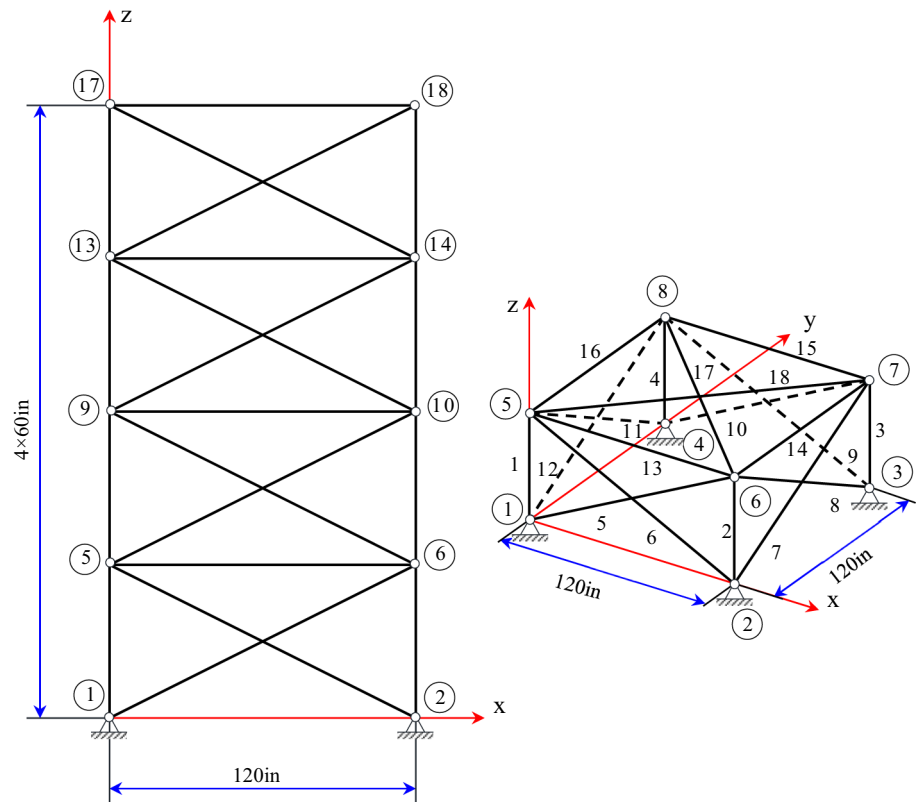


Table 11 Optimization results obtained for the 72-bar space truss with displacement and stress constraints

Design variables	HS	BB-BC	HPSO	HBB-BC	TLBO	This study	
						DE	DUL
A_i (in ²)	[16]	[73]	[74]	[3]	[75]		
A_1 - A_4	1.790	1.8577	1.857	1.9042	1.9064	1.8356	1.8606
A_5 - A_{12}	0.521	0.5059	0.505	0.5162	0.5061	0.5359	0.5008
A_{13} - A_{16}	0.100	0.1000	0.100	0.1000	0.1000	0.1002	0.1015
A_{17} - A_{18}	0.100	0.1000	0.100	0.1000	0.1000	0.1001	0.1011
A_{19} - A_{22}	1.229	1.2476	1.255	1.2582	1.2617	1.2991	1.2635
A_{23} - A_{30}	0.522	0.5269	0.503	0.5035	0.5111	0.4959	0.5060
A_{31} - A_{34}	0.100	0.1000	0.100	0.1000	0.1000	0.1001	0.1013
A_{35} - A_{36}	0.100	0.1012	0.100	0.1000	0.1000	0.1007	0.1009
A_{37} - A_{40}	0.517	0.5209	0.496	0.5178	0.5317	0.4759	0.4971
A_{41} - A_{48}	0.504	0.5172	0.506	0.5214	0.5159	0.5140	0.5077
A_{49} - A_{52}	0.100	0.1004	0.100	0.1000	0.1000	0.1000	0.1017
A_{53} - A_{54}	0.101	0.1005	0.100	0.1007	0.1000	0.1043	0.1032
A_{55} - A_{58}	0.156	0.1565	0.100	0.1566	0.1562	0.1002	0.1002
A_{59} - A_{66}	0.547	0.5507	0.524	0.5421	0.5493	0.4932	0.5185
A_{67} - A_{70}	0.442	0.3922	0.400	0.4132	0.4097	0.3840	0.4013
A_{71} - A_{72}	0.590	0.5922	0.534	0.5756	0.5698	0.5658	0.5374
Best weight (lb)	379.27	379.85	369.65	379.66	379.63	370.30	370.04
Weight error (%)	2.421	2.578	0.176	2.527	2.519	0.070	-
Constraint tolerance (%)	0.218	None	39.075	None	None	None	None
Number of FEAs	20,000	19,621	125,000	13,200	19,709	8580	1000
Total times (s)	-	-	-	-	-	488.135	63.715

Bold used to emphasize the best minimum weight design

Fig. 14 The weight convergence histories obtained using the DUL and DE for the 72-bars space truss

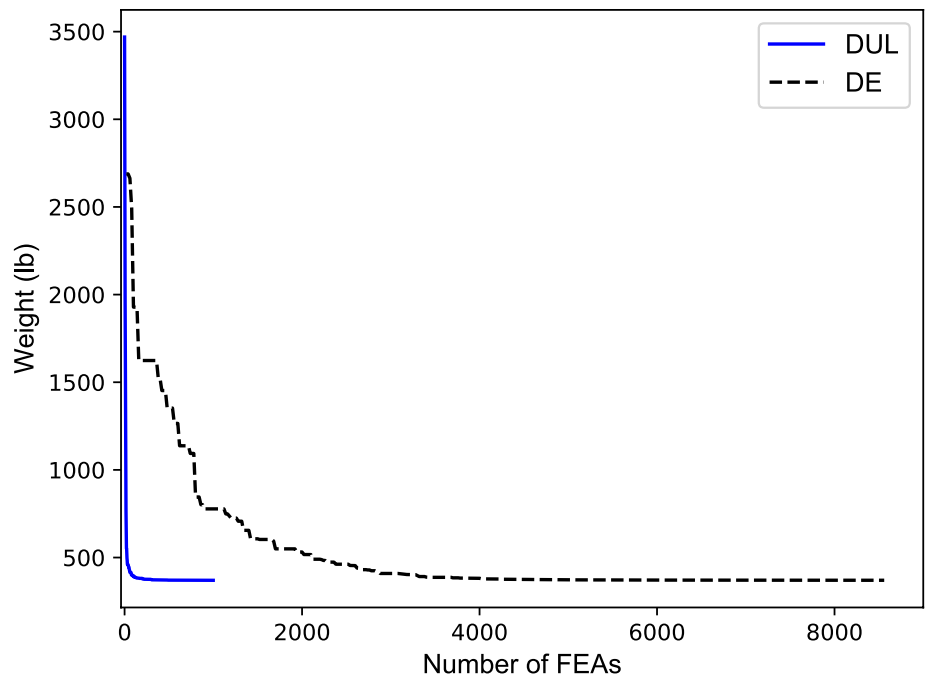


Fig. 15 A 200-bar planar truss structure

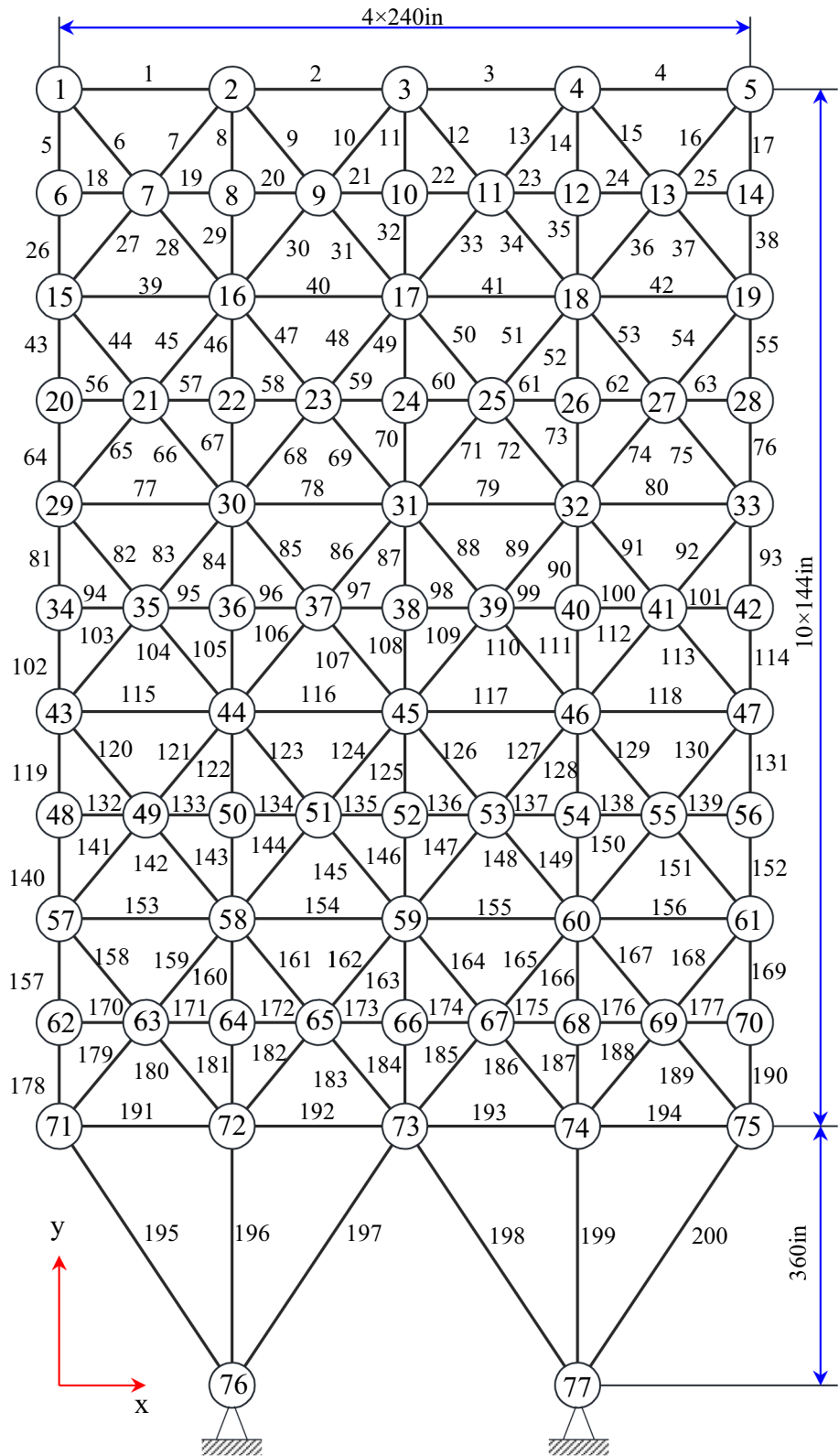


Table 12 Optimization results obtained for the 200-bar planar truss with stress constraints

Design variables	HS	HBB-BC	CMLPSA	EHS	TLBO	MCOA	This study	
	[16]	[3]	[76]	[19]	[75]	[15]	DE	DUL
A_i (in ²)								
1	0.1253	0.1033	0.1468	0.150	0.146	0.1390	0.1206	0.1183
2	1.0157	0.9184	0.9400	0.946	0.941	0.9355	0.9345	0.9907
3	0.1069	0.1202	0.1000	0.101	0.100	0.1000	0.1168	0.1142
4	0.1096	0.1009	0.1000	0.100	0.101	0.1000	0.1000	0.2323
5	1.9369	1.8664	1.9400	1.945	1.941	1.9355	1.9292	1.9579
6	0.2686	0.2826	0.2962	0.296	0.296	0.2909	0.2870	0.2906
7	0.1042	0.1000	0.1000	0.100	0.100	0.1000	0.1102	0.1624
8	2.9731	2.9683	3.1042	3.161	3.121	3.0816	3.0780	3.1520
9	0.1309	0.1000	0.1000	0.102	0.100	0.1000	0.2074	0.1380
10	4.1831	3.9456	4.1042	4.199	4.173	4.0816	4.0783	4.1834
11	0.3967	0.3742	0.4034	0.401	0.401	0.3967	0.4329	0.3843
12	0.4416	0.4501	0.1912	0.181	0.181	0.2959	0.1546	0.2115
13	5.1873	4.9603	5.4284	5.431	5.423	5.3854	5.3500	5.4466
14	0.1912	1.0738	0.1000	0.100	0.100	0.1000	0.1027	0.1272
15	6.2410	5.9785	6.4284	6.428	6.422	6.3853	6.3502	6.4490
16	0.6994	0.7863	0.5734	0.571	0.571	0.6332	0.5636	0.5317
17	0.1158	0.7374	0.1327	0.156	0.156	0.1842	0.5160	0.2149
18	7.7643	7.3809	7.9717	7.961	7.958	8.0396	7.9508	8.0113
19	0.1000	0.6674	0.1000	0.100	0.100	0.1000	0.1017	0.1434
20	8.8279	8.3000	8.9717	8.959	8.958	9.0395	8.9503	8.9967
21	0.6986	1.1967	0.7049	0.722	0.720	0.7460	0.8932	0.7054
22	1.5563	1.0000	0.4196	0.491	0.478	0.1306	0.1525	0.2450
23	10.9806	10.8262	10.8636	10.909	10.897	10.9114	11.0423	10.8275
24	0.1317	0.1000	0.1000	0.101	0.100	0.1000	0.1000	0.1265
25	12.1492	11.6976	11.8606	11.985	11.897	11.9114	12.0423	11.8557
26	1.6373	1.3880	1.0339	1.084	1.080	0.8627	0.9196	0.8580
27	5.0032	4.9523	6.6818	6.464	6.462	6.9169	6.7136	6.8944
28	9.3545	8.8000	10.8113	10.802	10.799	10.9674	10.7305	11.1690
29	15.0919	14.6645	13.8404	13.936	13.922	13.6742	13.8833	13.6032
Best weight (lb)	25447.10	25156.50	25445.63	25542.50	25488.15	25,450.18	25564.99	25547.90
Weight error (%)	0.012	1.154	0.018	0.363	0.149	–	0.451	0.384
Constraint tolerance (%)	3.69	9.97	0.071	None	None	None	None	None
Number of FEAs	48,000	9875	9650	22,851	28,059	27,720	47,100	4000
Total times (s)	–	–	–	–	–	–	3553.642	1305.273

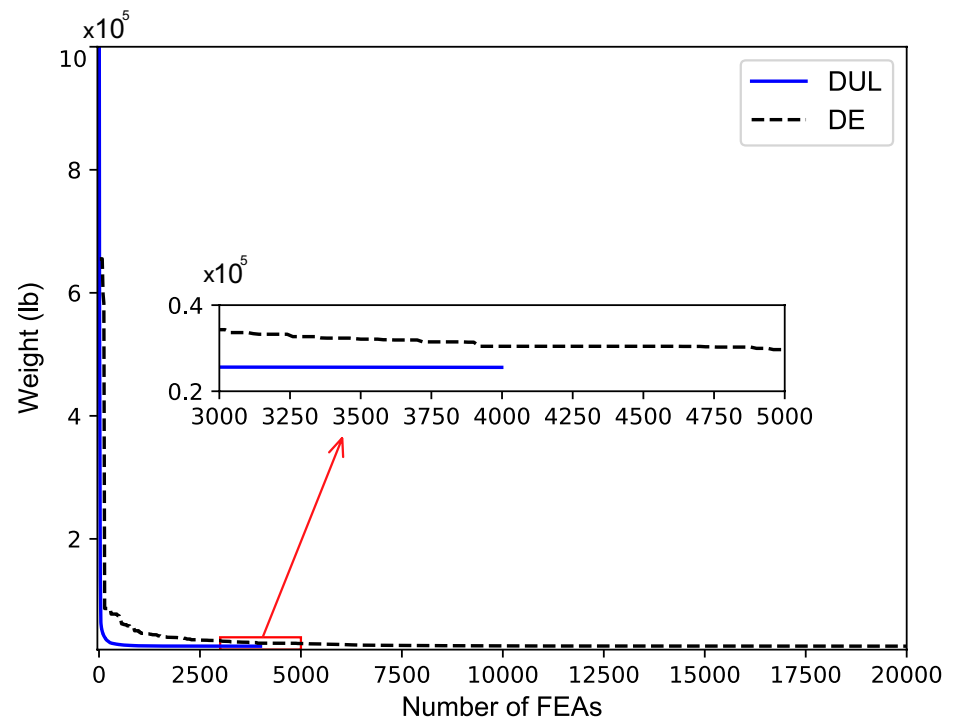
Bold used to emphasize the best minimum weight design

studies [15, 75]. However, this is demonstrated in this study by the DE algorithm. Despite the same parameters, there is a good agreement between the results obtained by the DE and other studies for the first six examples, but it is more difficult to resolve the last example. Herein, the optimum weight obtained by the DE (25564.99 lb) is much heavier than the MCOA (25,450.18 lb).

From Table 12, it is easily seen that the optimum weight gained by the DUL is larger than the MCOA [15] and

TLBO [75] with the error less than 0.4%. One explanation for these differences is that the backpropagation algorithm based on the gradient descent method deals with the initial value of the weights, biases, and learning rate [83]. Furthermore, the total computational cost of DUL (1305.273 s) reduces by more than a half of DE (3553.642 s). A comparison of the convergence rate between the DUL and DE is shown in Fig. 16. Obviously, the DUL always converges much more quickly than the DE. Again, this shows the efficiency of our method.

Fig. 16 The weight convergence histories obtained using the DUL and DE for the 200-bars planar truss



5 Conclusions

In this article, an efficient DUL-based methodology is first introduced to directly perform optimization of truss structures under multiple constraints. The members' cross-sectional areas are parameterized by the parameters of the DNN. According to that core idea, the weights and biases are considered as the design variables of the structural optimization problem, instead of the cross-sectional areas. The loss function is constructed relying on the output values of the DNN and the structural responses using FEA. The optimum weight of the structure is found as soon as the training phase ends without any other algorithms. The efficiency of the proposed approach is demonstrated through several numerical examples for size optimization of truss structures. The obtained results have indicated that the optimum weight obtained by this work is a good agreement with six of the seven tests. The DUL dramatically saves computational cost in almost all problems in comparison with other algorithms. In addition, its convergence speed is very fast at the beginning of iterations. This approach promises to extend its applications to more complex optimization problems such as discrete variables, multi-objective, and so on.

However, the implementation of this study may encounter some challenges which have still left unsolved. Firstly, the optimal architecture and hyperparameters of the DNN need to be indicated for each research area. In this study, GS and trial-error-tuning approaches were used to choose the best-fitted model, but it depends on prior knowledge and

expert experience too much. To circumvent this limitation, one of the promising directions is recommended as genetic algorithms, Bayesian optimization techniques, etc. Secondly, the examined optimization problems are non-convex ones, whilst Adam is a gradient-based optimization algorithm. Hence, optimized solutions may be trapped in local minimum due to initial parameters. Accordingly, other robust optimizers such as gradient descent with momentum, Nesterov accelerated gradient, etc. [84] are promising to circumvent these shortcomings.

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Declarations

Conflict of interest The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

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